



LIMITS OF STEADY PROPAGATION OF HYDROGEN DEFLAGRATIONS AND DETONATIONS

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Flame propagation in tubes



- **Lower limit** \Rightarrow LAMINAR FLAME (m/s)
- **Upper limit** \Rightarrow CJ DETONATION (km/s)
- **Between limits** \Rightarrow spectrum of TURBULENT FLAMES depending on:
 - Initial conditions: pressure, temperature, composition
 - Geometry: size, obstacles, etc.
- **Smooth tubes** \Rightarrow continuous flame acceleration and abrupt DDT
- **Rough (obstructed) tubes** \Rightarrow several distinct regimes of steady flame propagation



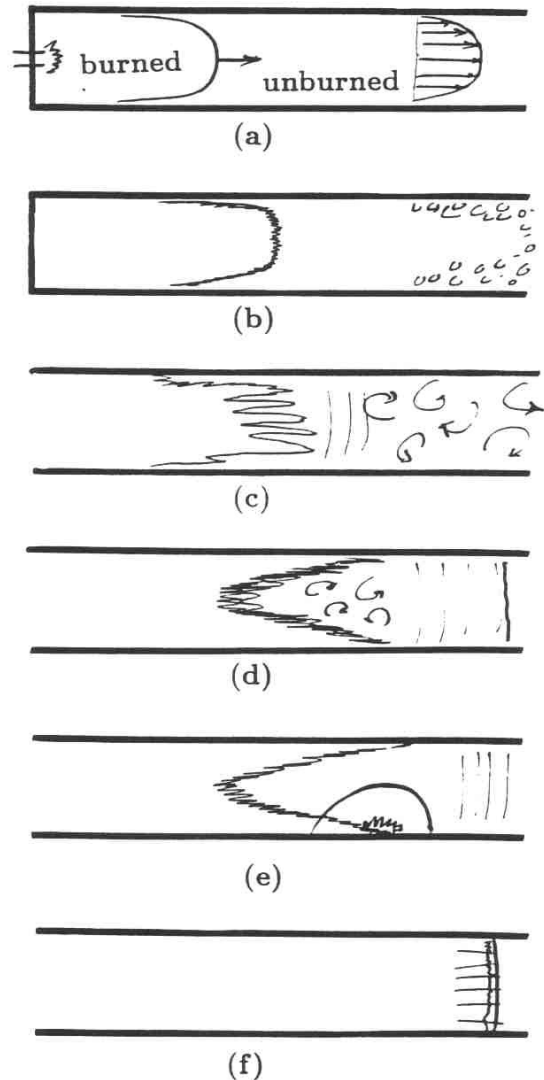
Experimental Composition Limits



Mixture	Deflagration lean limit [% fuel by vol.]	Detonation lean limit [% fuel by vol.]	Detonation rich limit [% fuel by vol.]	Deflagration rich limit [% fuel by vol.]
$H_2 - O_2$	4.6	15	90	93.9
$H_2 - \text{air}$	4	18.3	59	74
$CO - O_2$	15.5	38	90	93.9
$(CO+H_2)-O_2$	12.5	17.2	91	92
$(CO+H_2)-\text{air}$	6.05	19	59	71.8
$NH_3 - O_2$	13.5	25.4	75	79
$C_3H_8 - O_2$	2.4	3.2	37	55
$C_2H_2 - O_2$	2.8	3.5	92	93
$C_4H_{10}O - \text{air}$	1.85	2.8	4.5	36.5

Source:

Kuo, Principles of Combustion, 2005

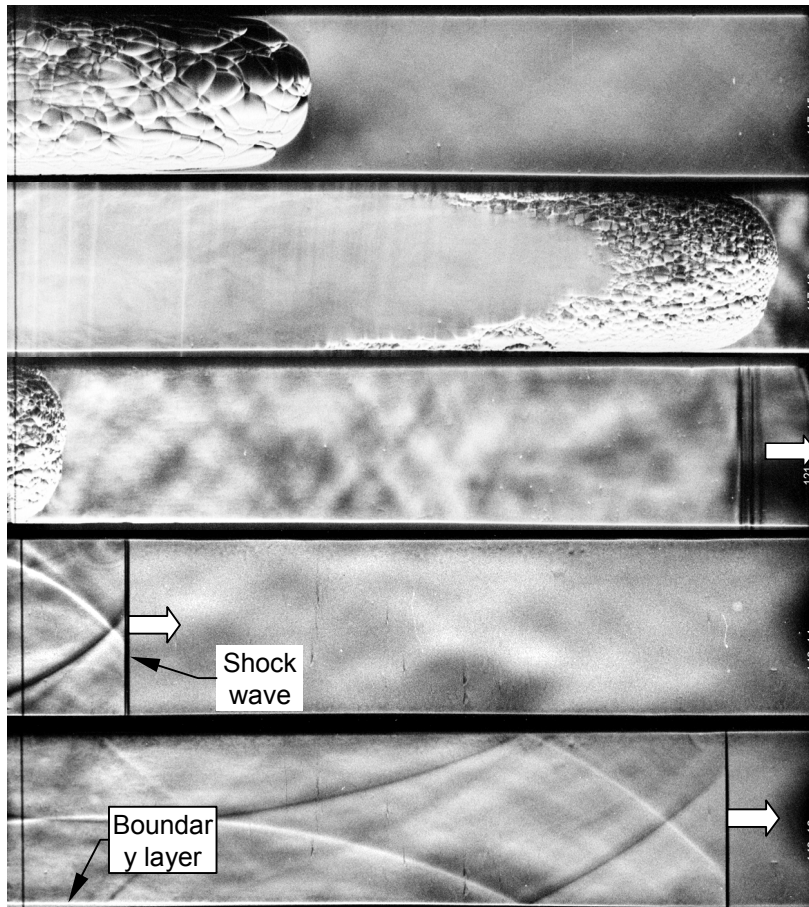


- the initial configuration showing a smooth flame and the laminar flow ahead;
- first wrinkling of flame and instability of the upstream flow;
- breakdown into turbulent flow and a corrugated flame;
- production of pressure waves ahead of the turbulent flame;
- local explosion of a vertical structure within the flame;
- transition to detonation.

(Shepherd&Lee, 1992)

Effect of boundary layer on the flame acceleration and DDT

Premixed flames in smooth closed tube - stoichiometric hydrogen-oxygen



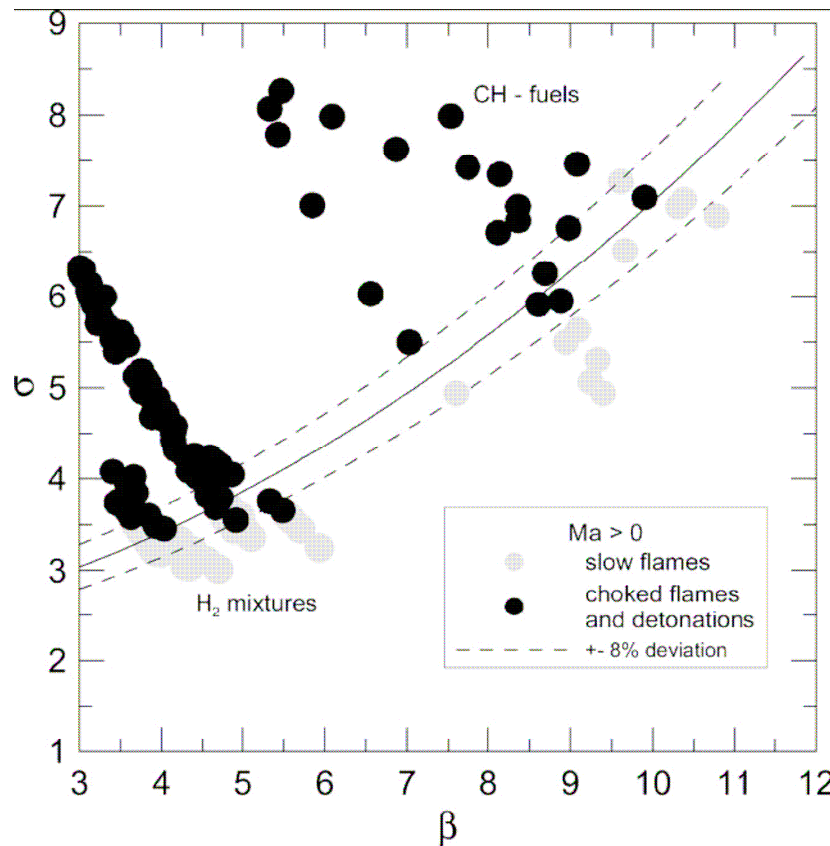
Shadow photograph of early stage of flame propagation

$p_0=0.75$ bar

at 210-440 mm from ignition

Ignition by electric spark of 20mJ

(Kuznetsov M., Dorofeev S., 2005)



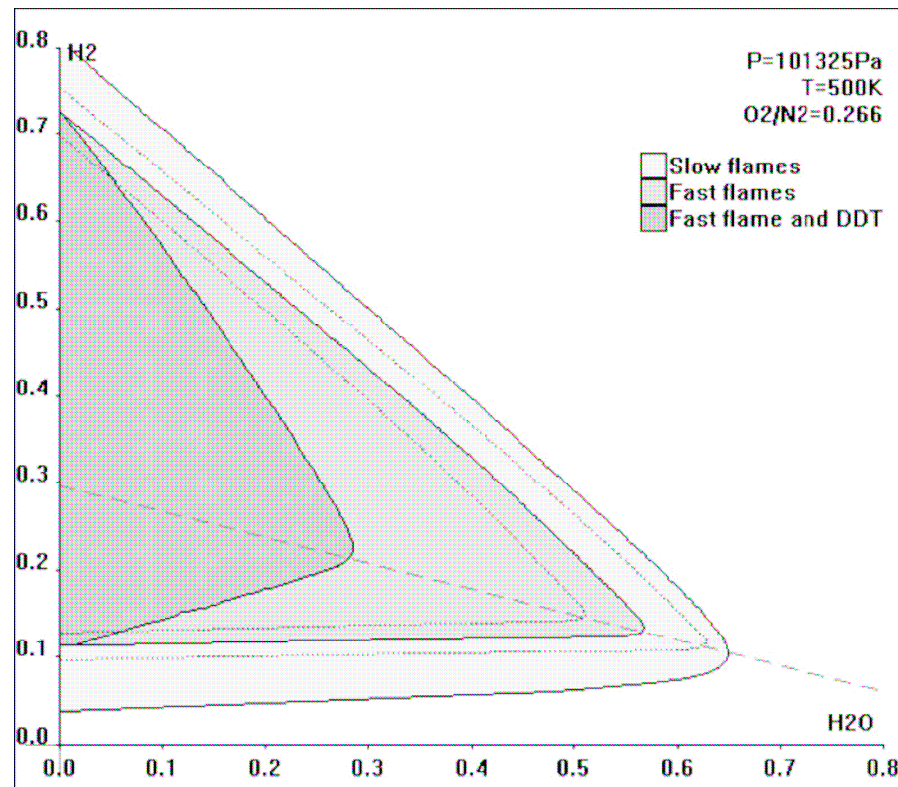
Zel'dovich number:

$$\beta = \frac{E_a (T_b - T_u)}{RT_b^2}$$

Expansion ratio:

$$\sigma = \frac{\rho_u}{\rho_b}$$

Source: S.Dorofeev et al., *Journal of Loss Prevention in the Process Industries* 14 (2001) 583–589



Explosion limits for H₂/air/H₂O mixtures at $T=500\text{ K}$ and $p=1\text{ atm}$. Range of uncertainty of fast flame boundary is shown by dotted lines

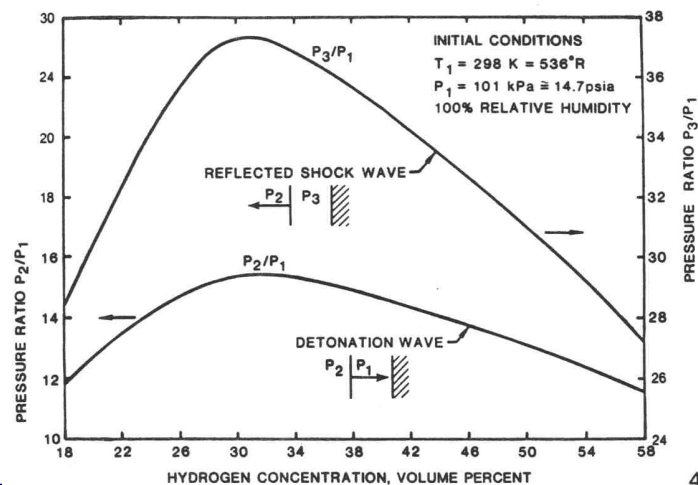
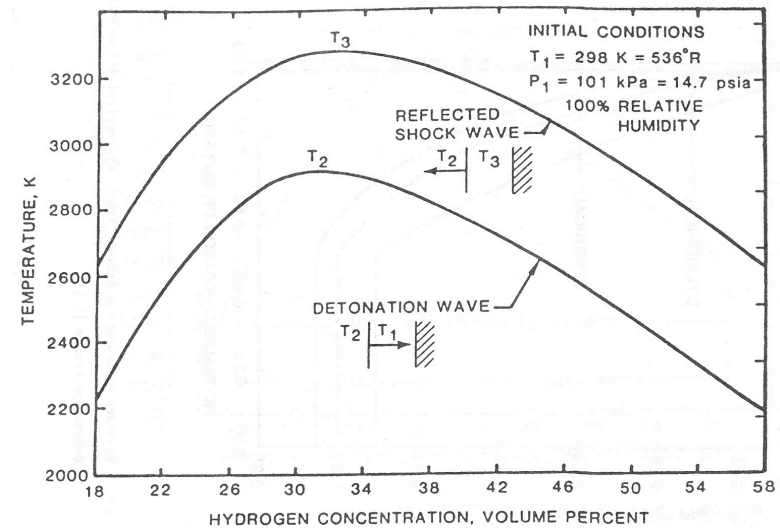
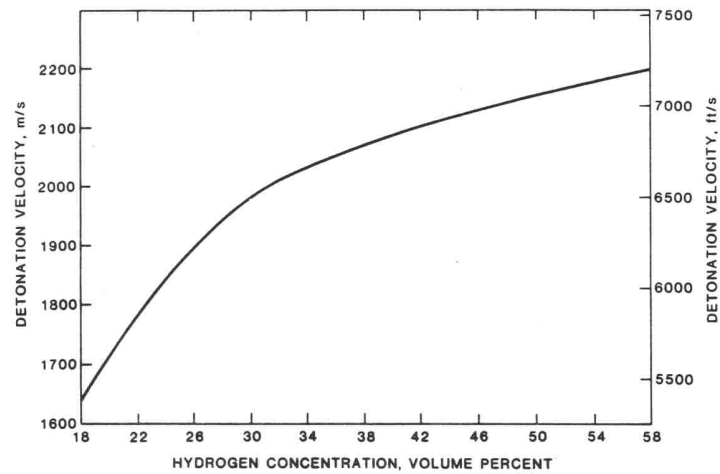
Source: S.Dorofeev et al., *Journal of Loss Prevention in the Process Industries* 14 (2001) 583–589



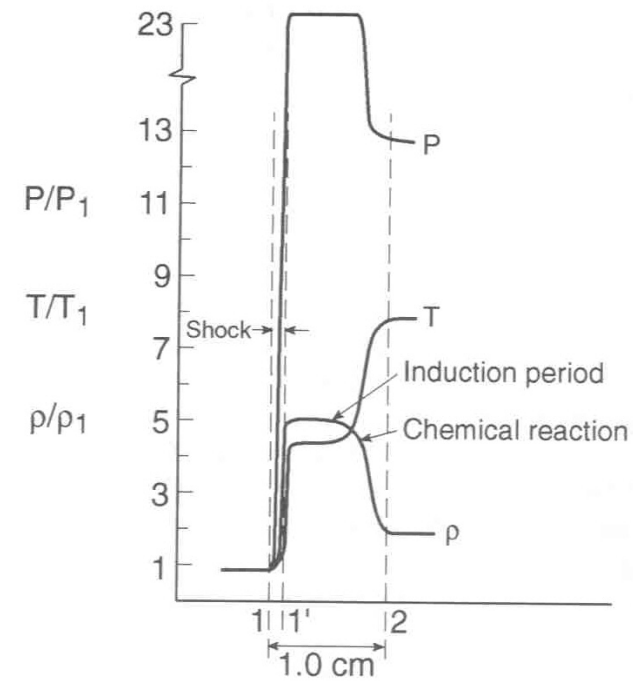
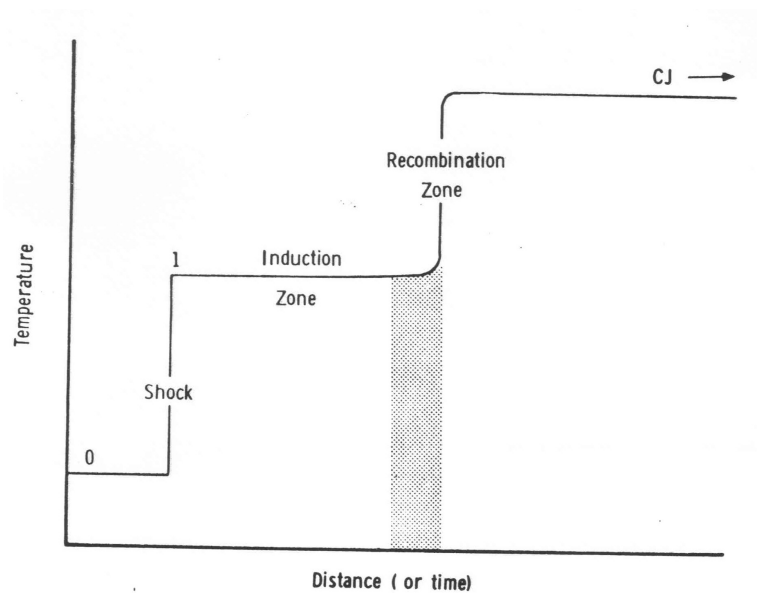
CJ Detonation



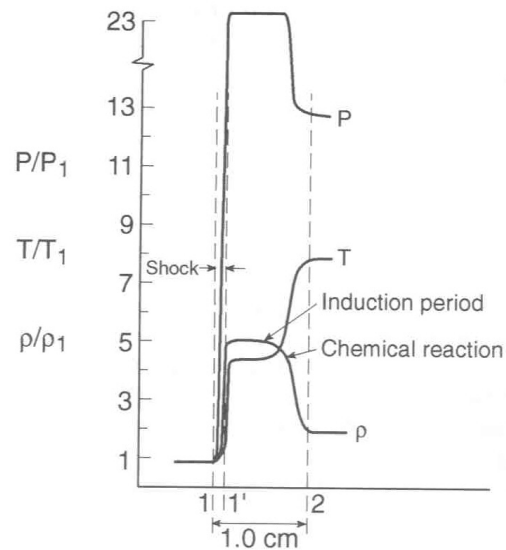
Hydrogen-Air CJ Detonation Parameters



- Velocity
 - Pressure
 - Temperature
- Are simple to calculate from equilibrium codes:
- NASA
STANJAN
SUPERSTATE

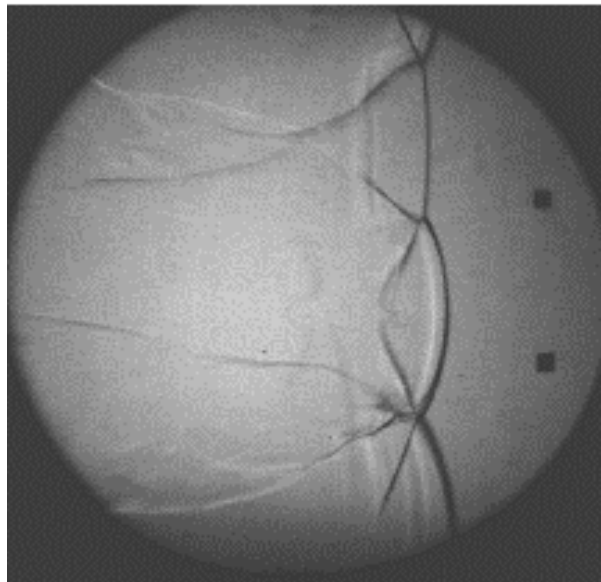
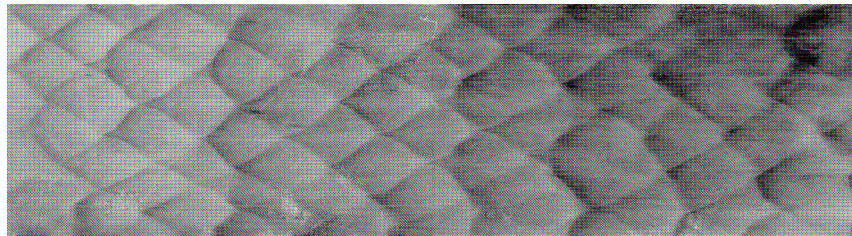
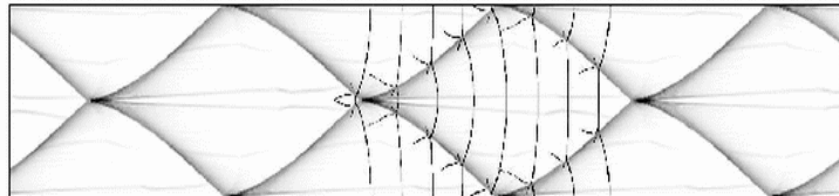


Calculated values of the physical parameters of ZND model for various hydrogen and propane detonations (Glassman I., Combustion, 1996)

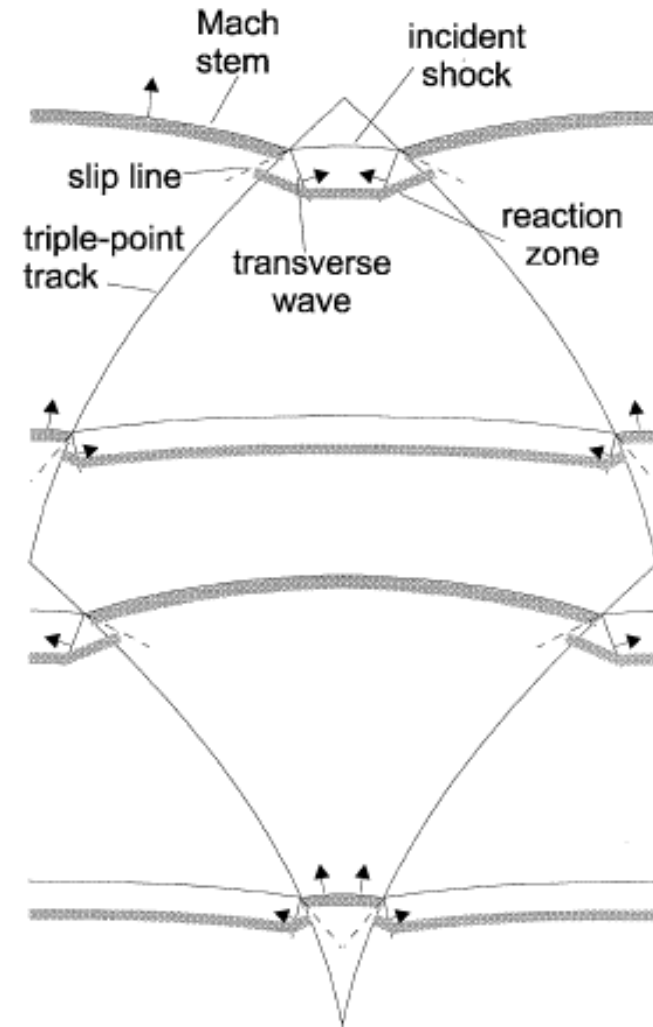


	1	1'	2
H2 – air ($\Phi = 1.2$)			
M	4.86	0.41	1.00
U [m/s]	2033	377	1129
P [bar]	1	28	16
T [K]	298	1546	2976
ρ/ρ_1	1.00	5.39	1.80
H2 – O2 ($\Phi = 1.1$)			
M	5.29	0.40	1.00
U [m/s]	2920	524	1589
P [bar]	1	33	19
T [K]	298	1773	3680
ρ/ρ_1	1.00	5.57	1.84

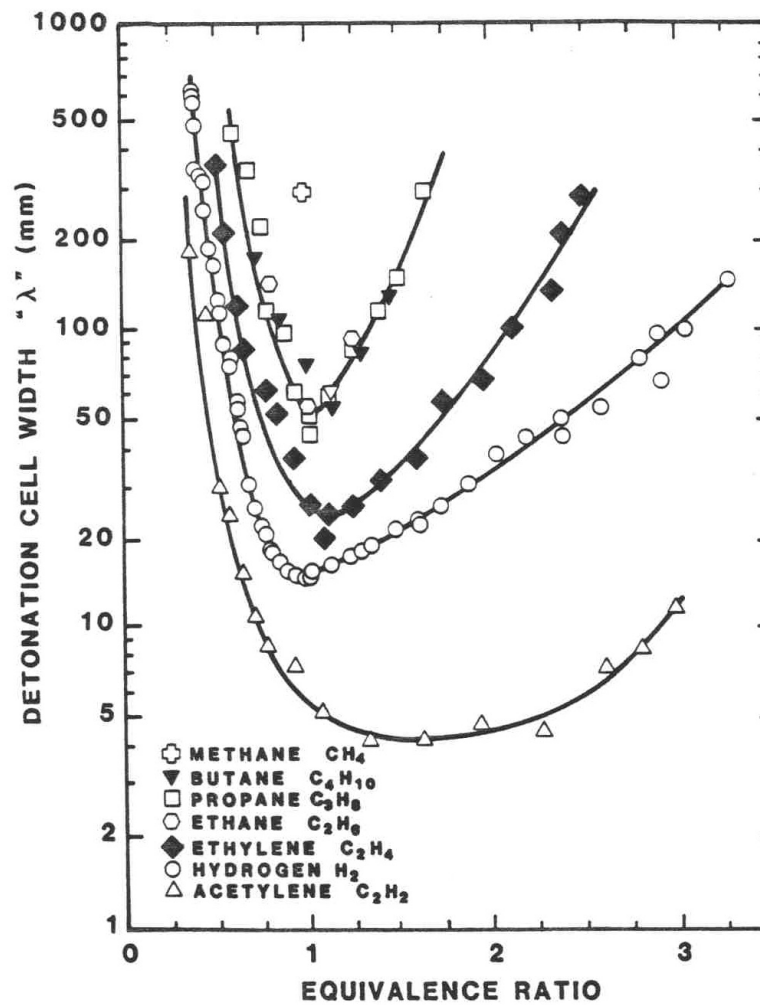
Detonation wave structure



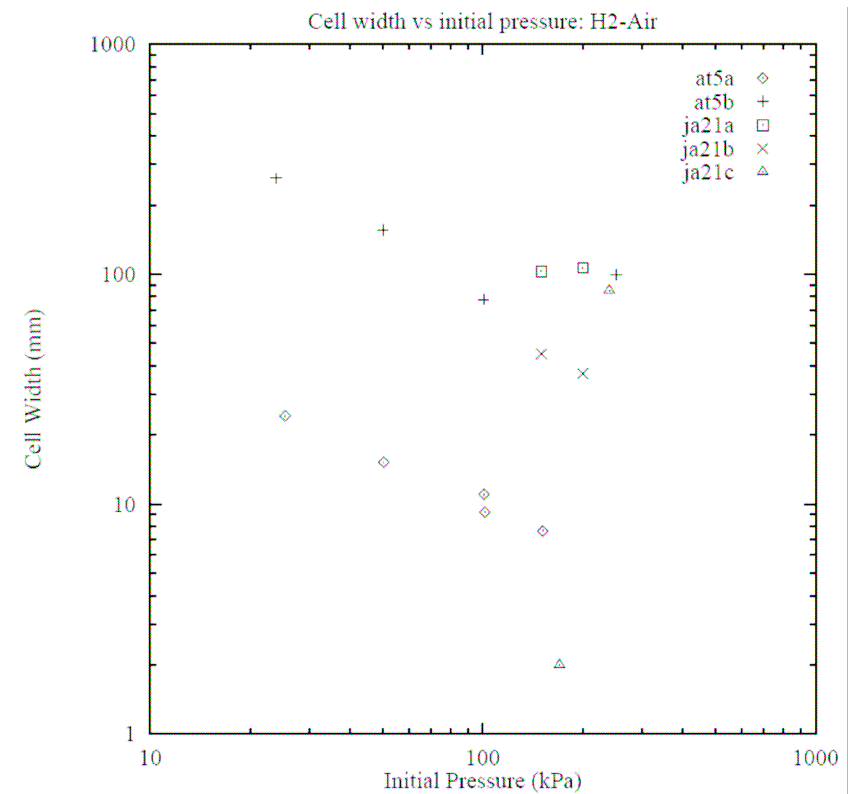
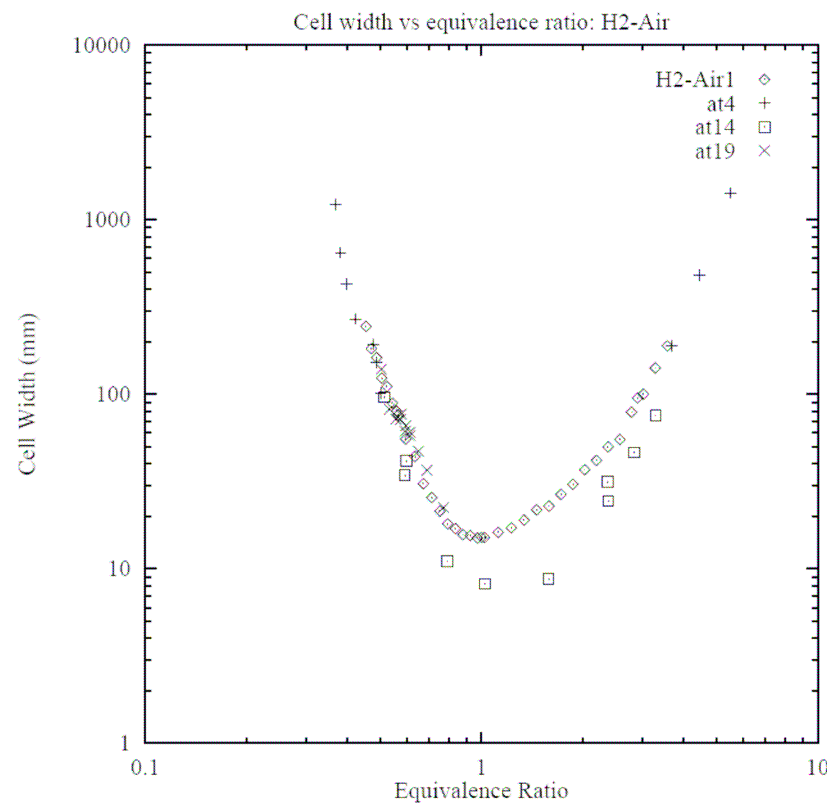
$2H_2+O_2+17Ar$ at 20kPa
(Austin&Shepherd)



Fuel-air mixtures



hydrogen-air mixtures

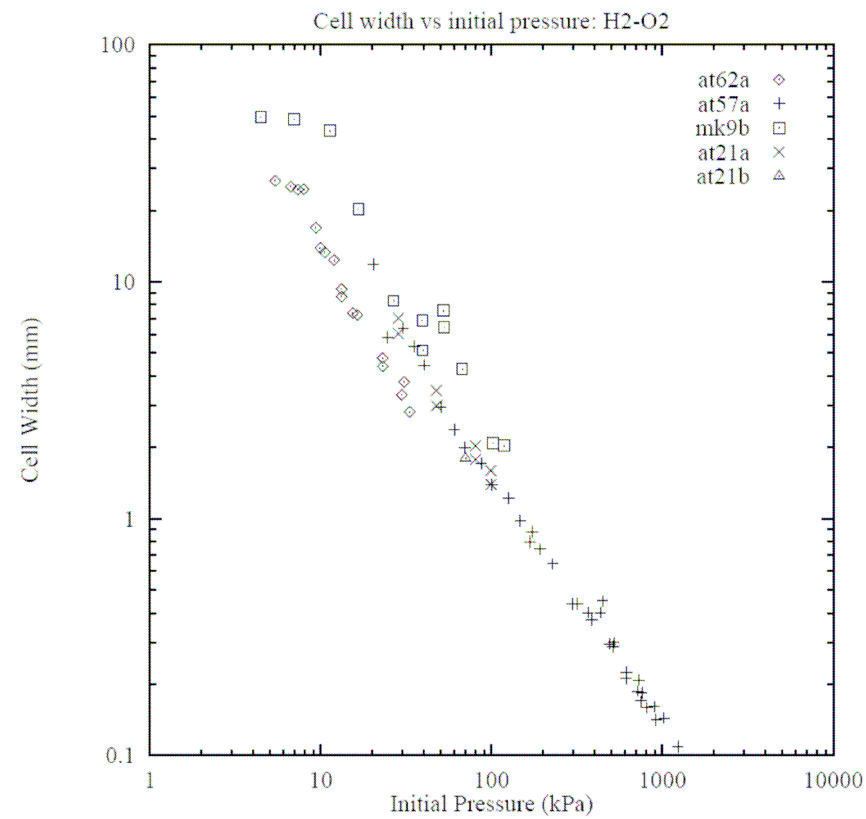




Detonation cell size



hydrogen-oxygen mixtures



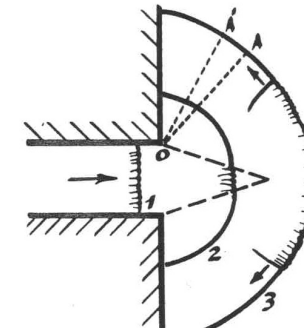
- propagation limit: $d_{\text{tube}} > d_f$

$$d_f = \lambda / \pi$$

- critical tube diameter for diffraction: $d_{\text{tube}} > d_c$

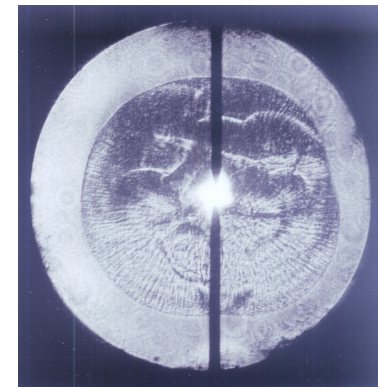
$$\text{Tube: } d_c = 13 \lambda$$

$$\text{Square channel: } l_c = 10 \lambda$$

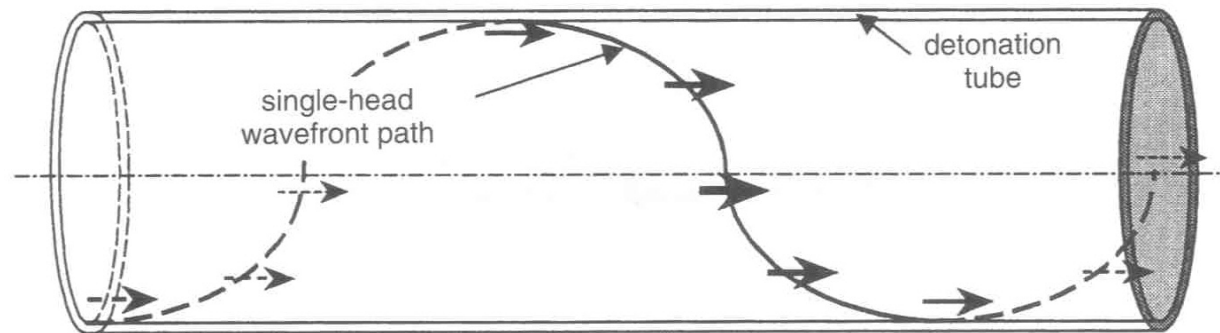


- Critical energy for direct initiation: $E > E_c$

$$E_c = 430 \rho_0 U_{CJ}^2 \lambda^3$$

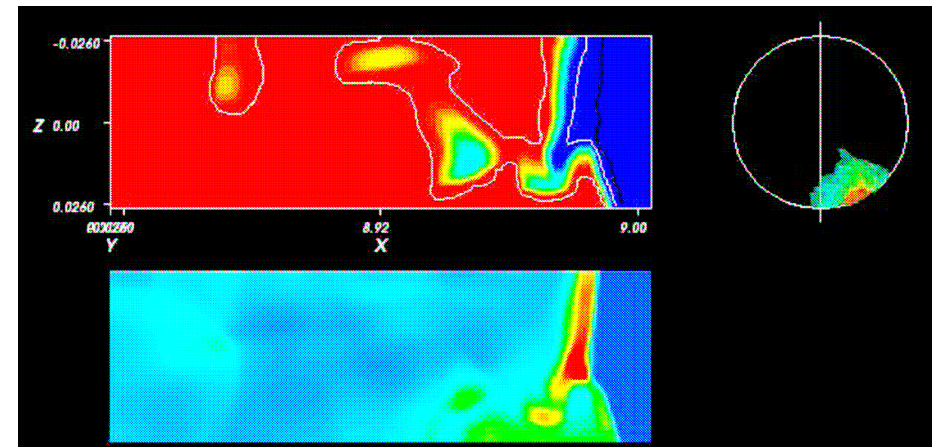
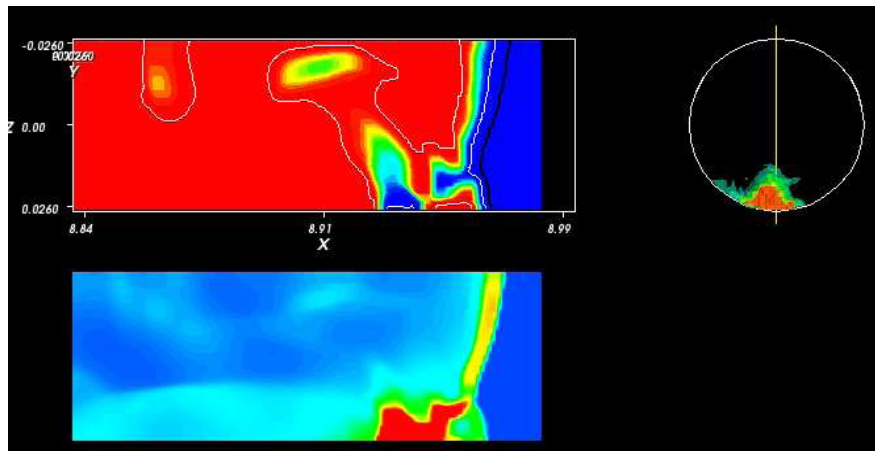


- single spin (head)





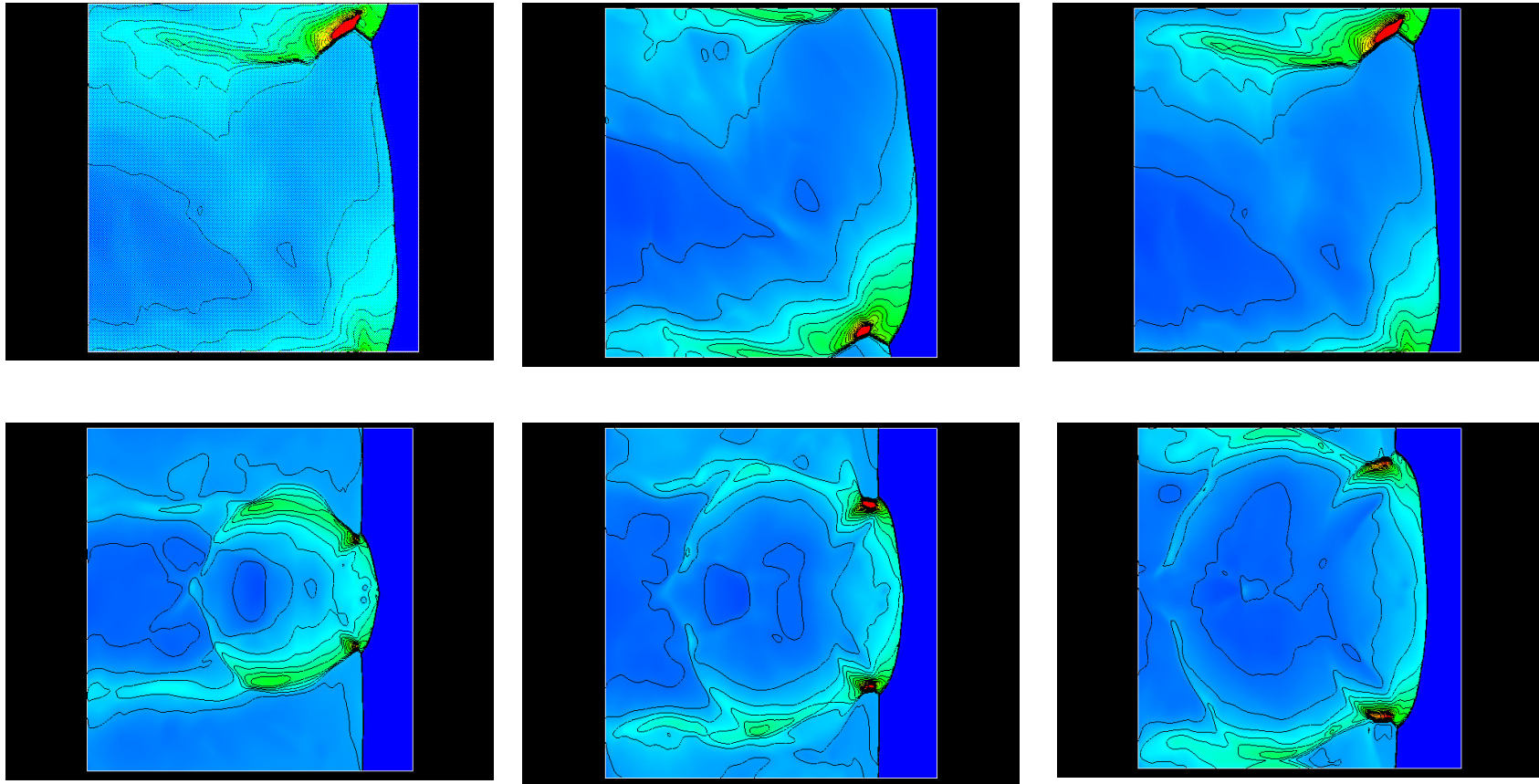
Experimental soot traces for CH₄ + 2O₂ mixture at P₀ = 50mbar.



CH₄/O₂ spinning detonation simulation

Source: F.Virot et al., 21st ICDERS, Poitiers, 2007

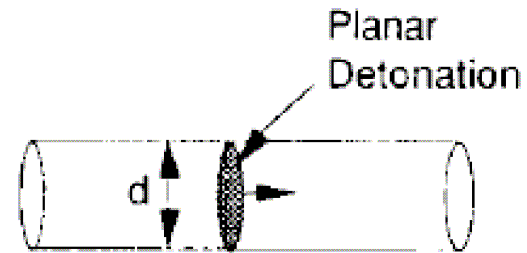
Single spin detonation



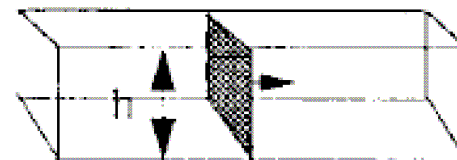
Comparison of pressure contours on the wall

Source: N.Tsuboi et al., 21st ICEDERS, Poitiers, 2007

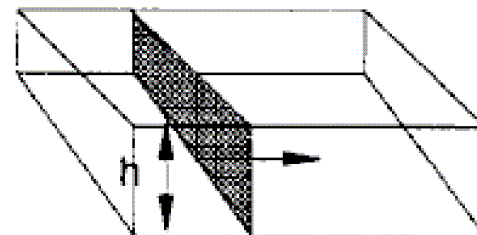
- propagation limit



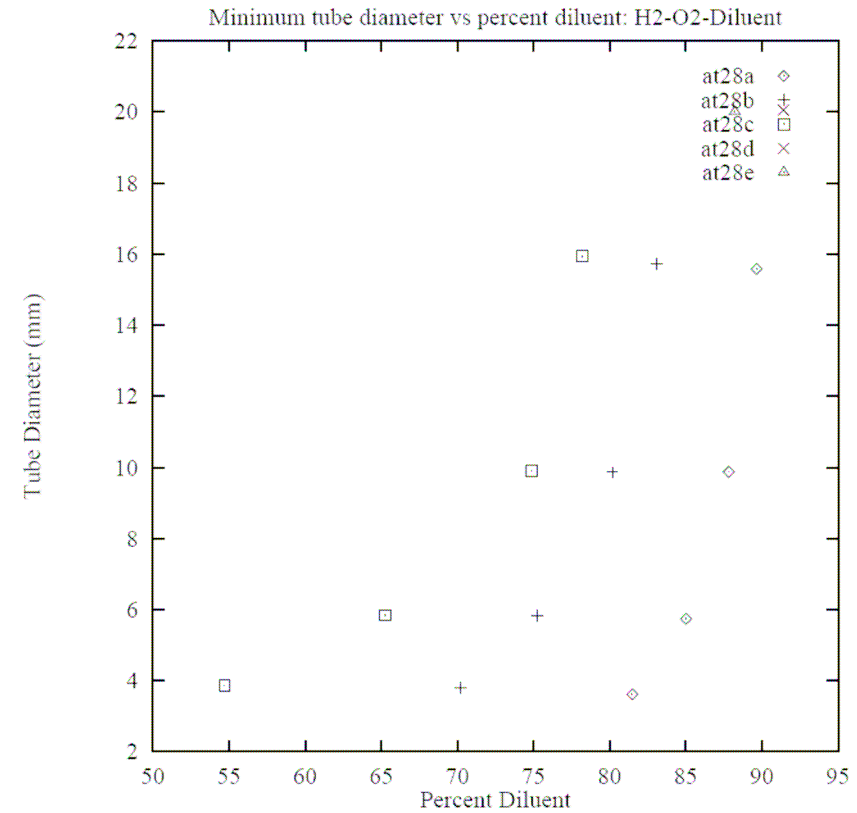
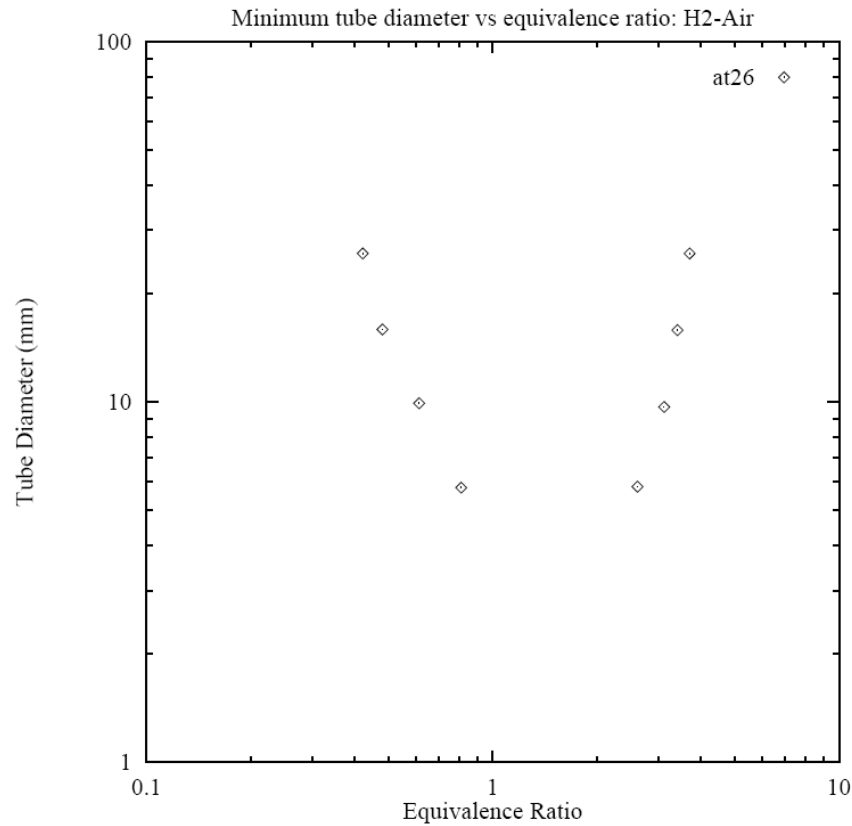
$$d > \lambda/3$$



$$h > \lambda$$

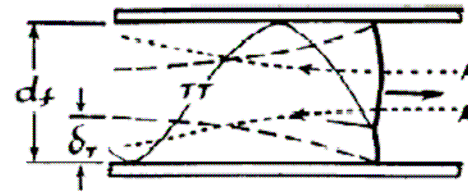


$$h > \lambda$$

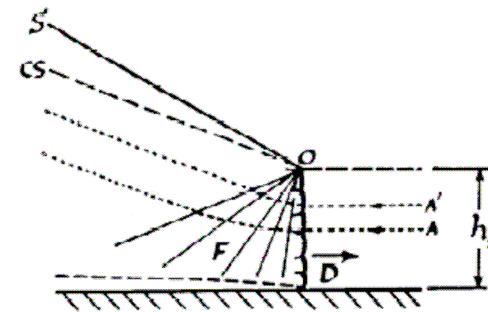


at28a: $T = 298\text{ K}$, $P = 100\text{ kPa}$, $\phi = 1$, 80-90% Ar; **at28b:** $T = 298\text{ K}$, $P = 100\text{ kPa}$, $\phi = 1$, 70-80% He; **at28c:** $T = 298\text{ K}$, $P = 100\text{ kPa}$, $\phi = 1$, 55-75% N₂; **at28d:** $T = 298\text{ K}$, $P = 100\text{ kPa}$, $\phi = 1$, 90% Ar; **at28e:** $T = 298\text{ K}$, $P = 100\text{ kPa}$, $\phi = 1$, 86% He

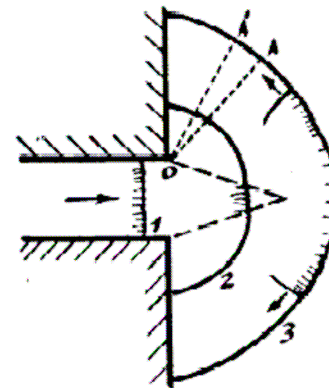
- Confined



- Unconfined

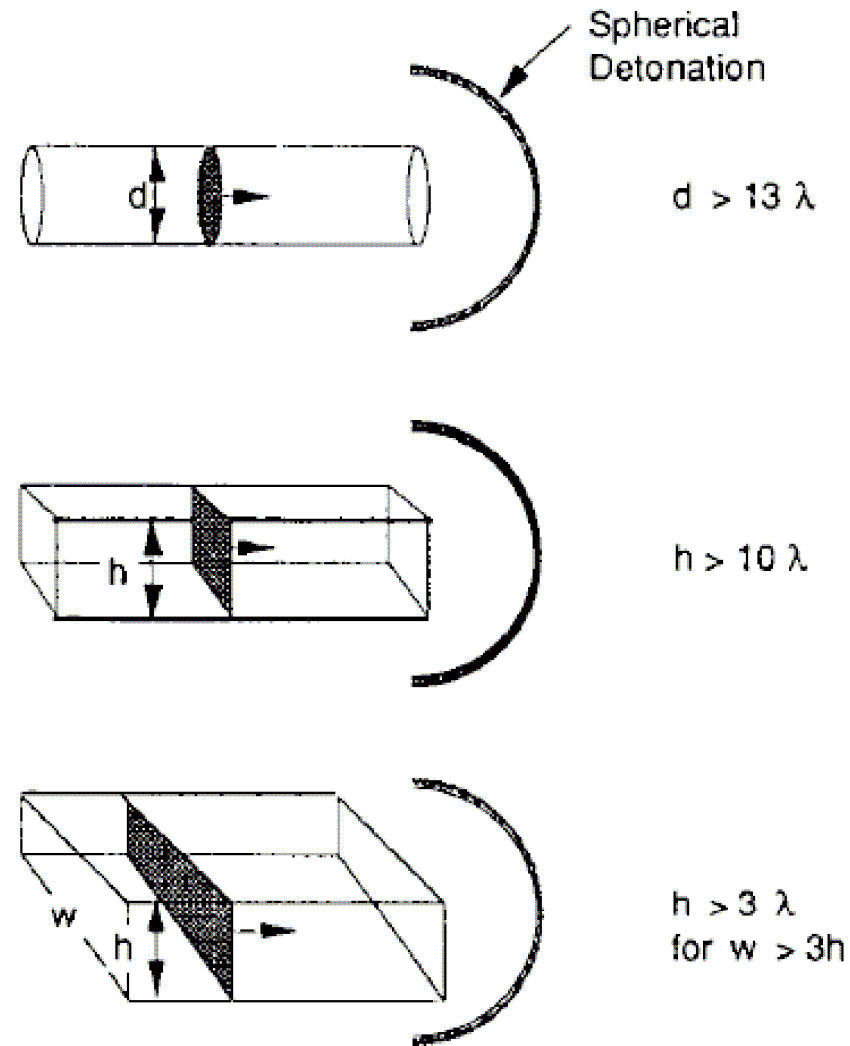


- Diffraction



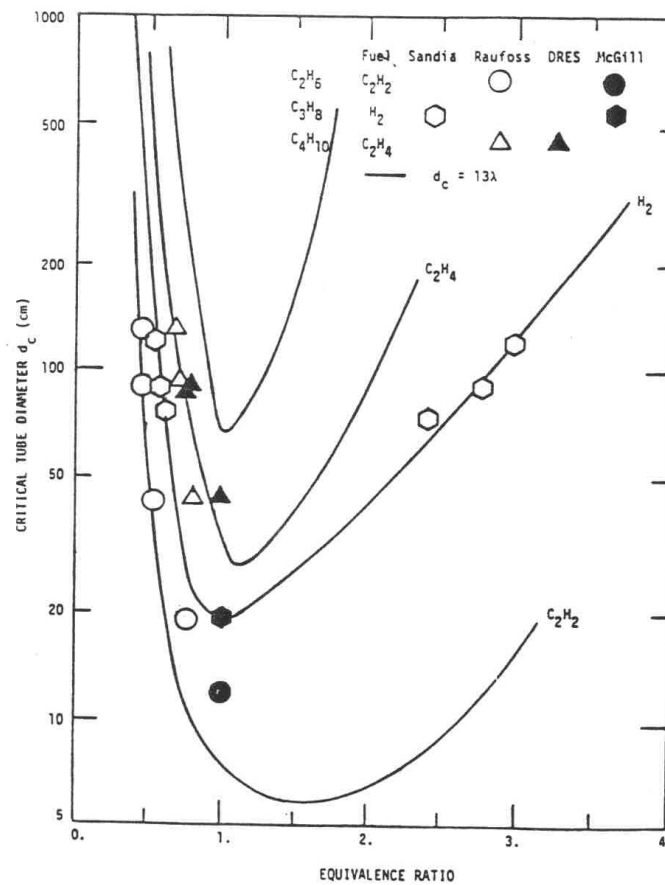
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- critical tube diameter for diffraction to unconfined space



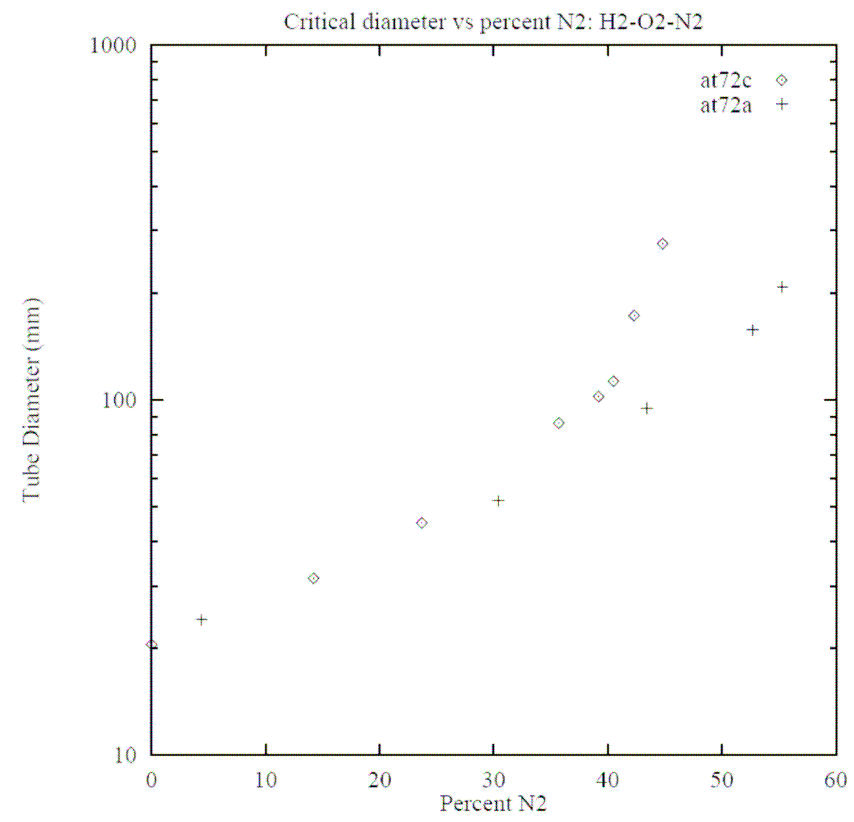
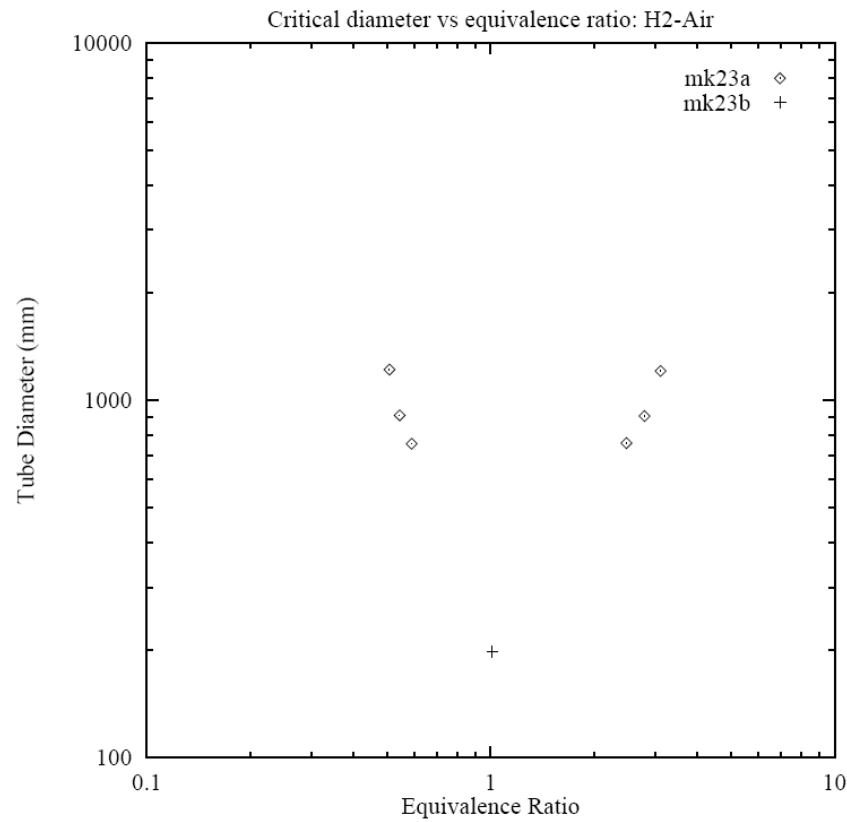
Critical tube diameter

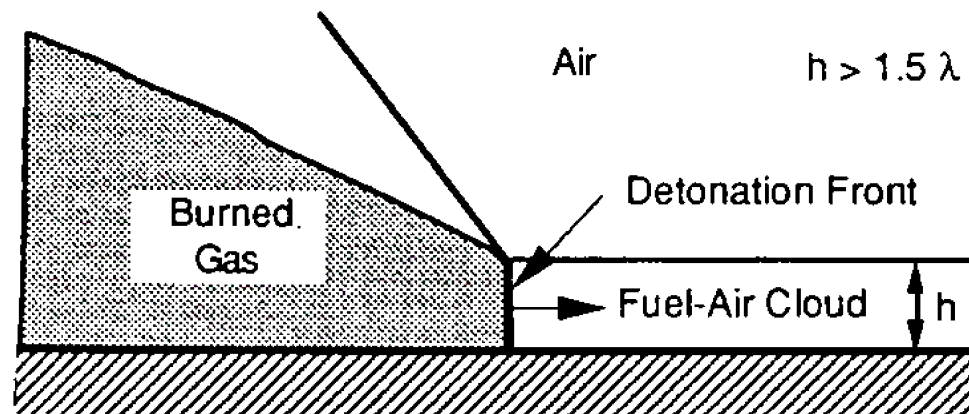
Fuel-Air Mixtures ($T_o = 300\text{ K}$, $P_o = 1\text{ atm}$)



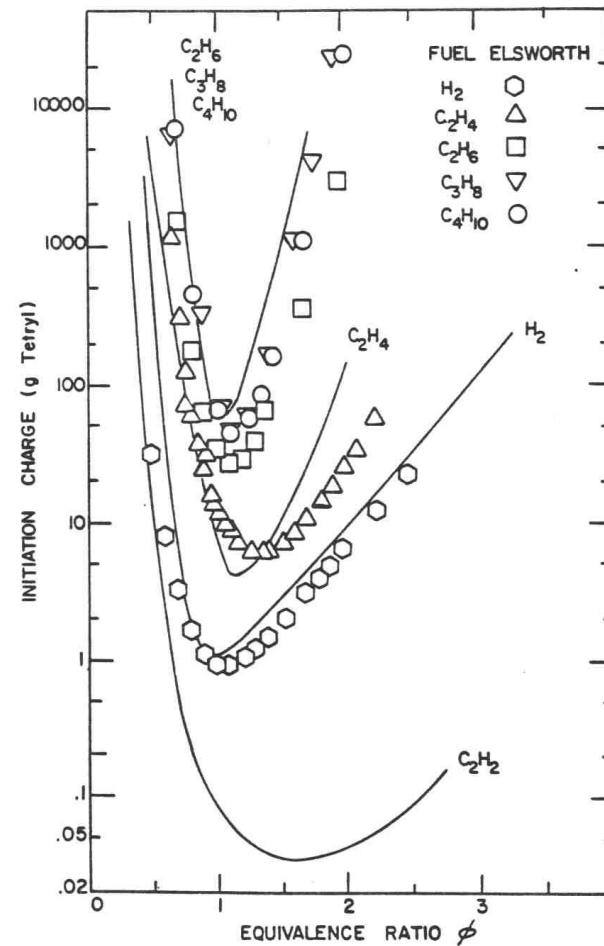


Critical tube diameter





Fuel-Air Mixtures ($T_o = 300$ K, $P_o = 1$ atm)





Detonation database



Detonation Database

[Accessing the Data](#) | [Summary Graphs](#) | [Data Sets](#) | [References](#) | [Database Search](#)

Abstract

Welcome to the GALCIT Explosion Dynamics Laboratory Detonation Database. The goal of this project is to compile, catalog and present experimental data on gaseous detonations. These data currently include cell width, critical tube diameter, initiation energy, and minimum tube diameter. They are formatted in tables and summary graphs, with citations to the original references. A printed version and a World Wide Web version have been prepared. The purpose of this database is to facilitate explosion hazards evaluations and comparisons with numerical simulations of detonation behavior.

[Introduction](#) to the Detonation Database project.

[Contributors](#). Authors of the database.

[Disclaimer](#). We're not perfect.

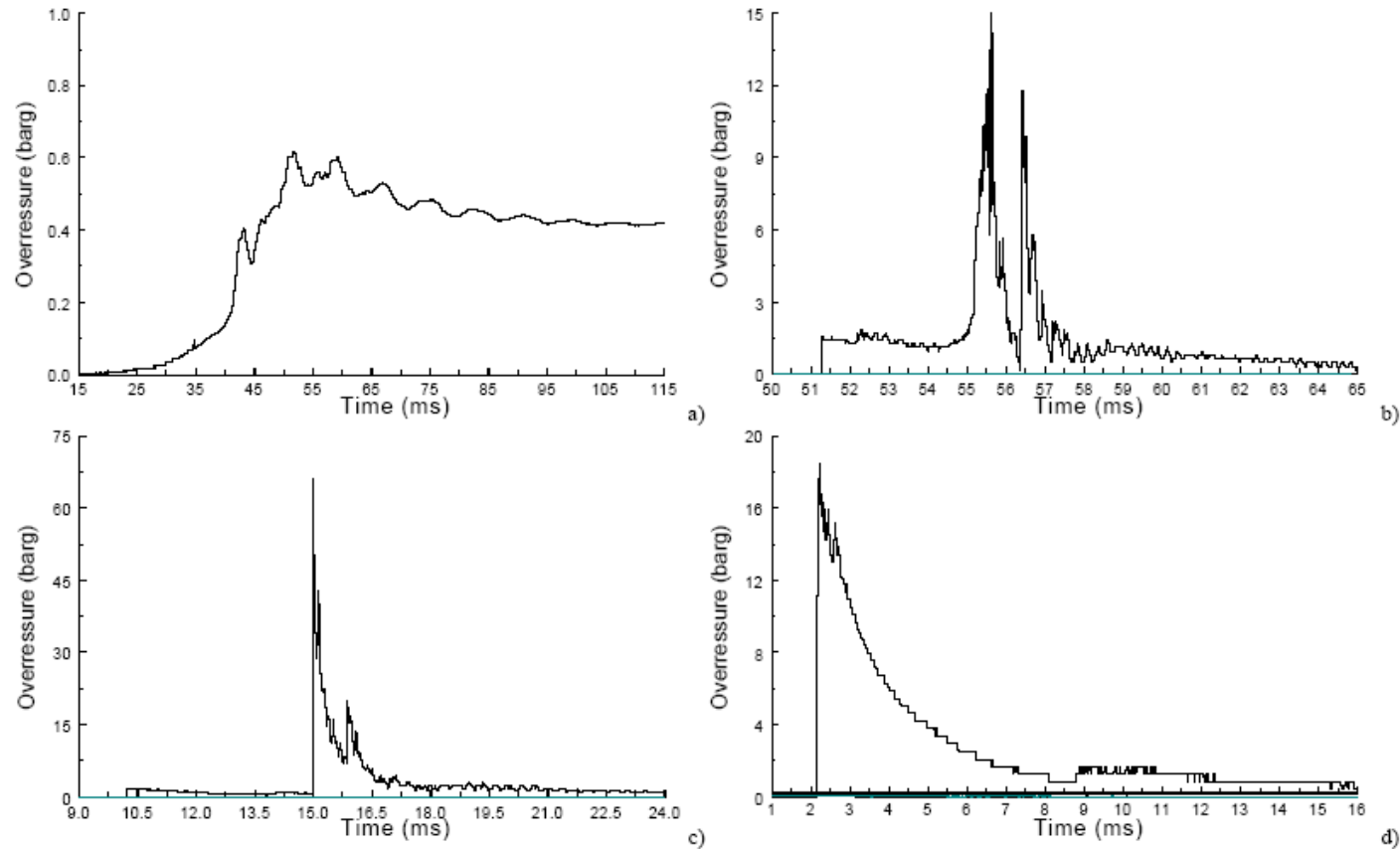
[Citations](#). Using the data in publications.

[How to Access the Data](#). Some useful information and tips.

[How the Database Works](#). For those who are interested.

[The Database](#). Links to the different branches.

Edited Last: Jan 29, 2005
Joe Shepherd



a) Slow deflagration; b) fast deflagration; c) overdriven detonation DDT; d) CJ detonation



Transition distance to DDT



Depends on:

- Combustible mixture (chemistry and thermodynamics)
- Tube diameter - for hydrogen-air in smooth tube:
 - 8 m in 50 mm tube
 - 30 m in 400 mm tube
- Ignition source
- Obstacles, wall roughness
- Initial conditions
- ???

Flame velocity versus fuel concentration for H₂-air mixtures

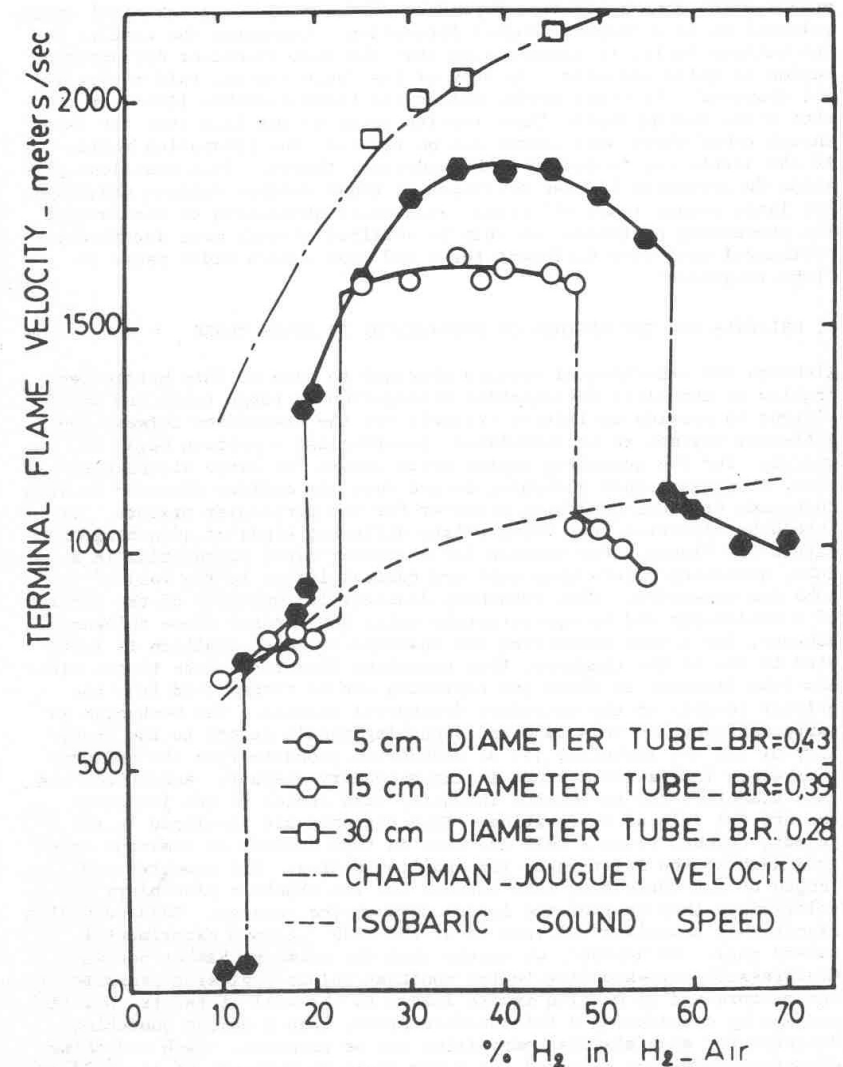
10 m long tubes of 5 cm, 15 cm and 30 cm in internal diameter with obstacles (orifice plates).

$BR = 1 - d^2/D^2$ – blockage ratio

d - orifice diameter

D - tube diameter

(Lee, 1986)

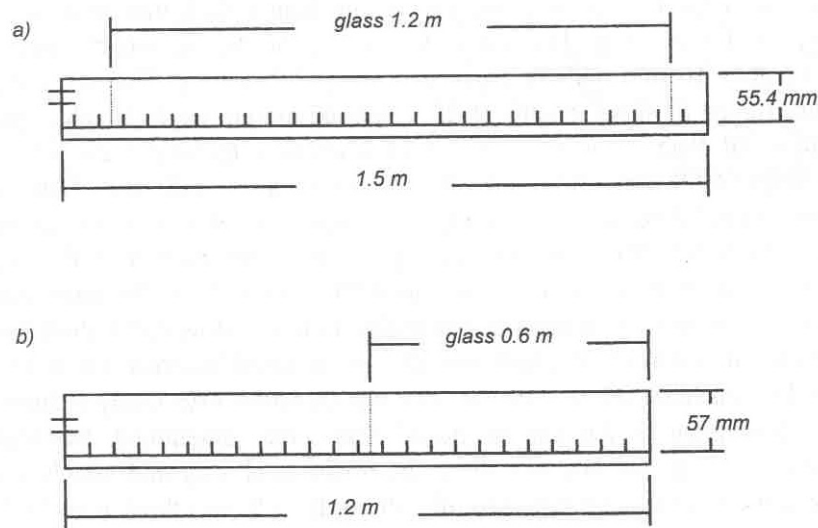




Regimes of flame propagation in tubes with obstacles

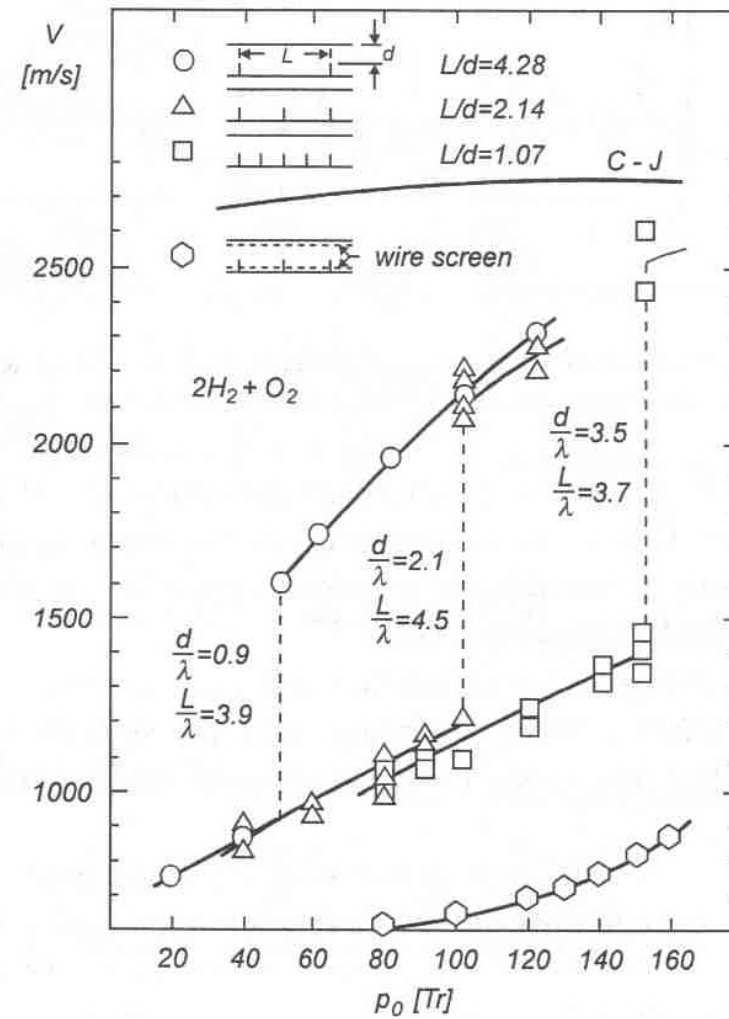


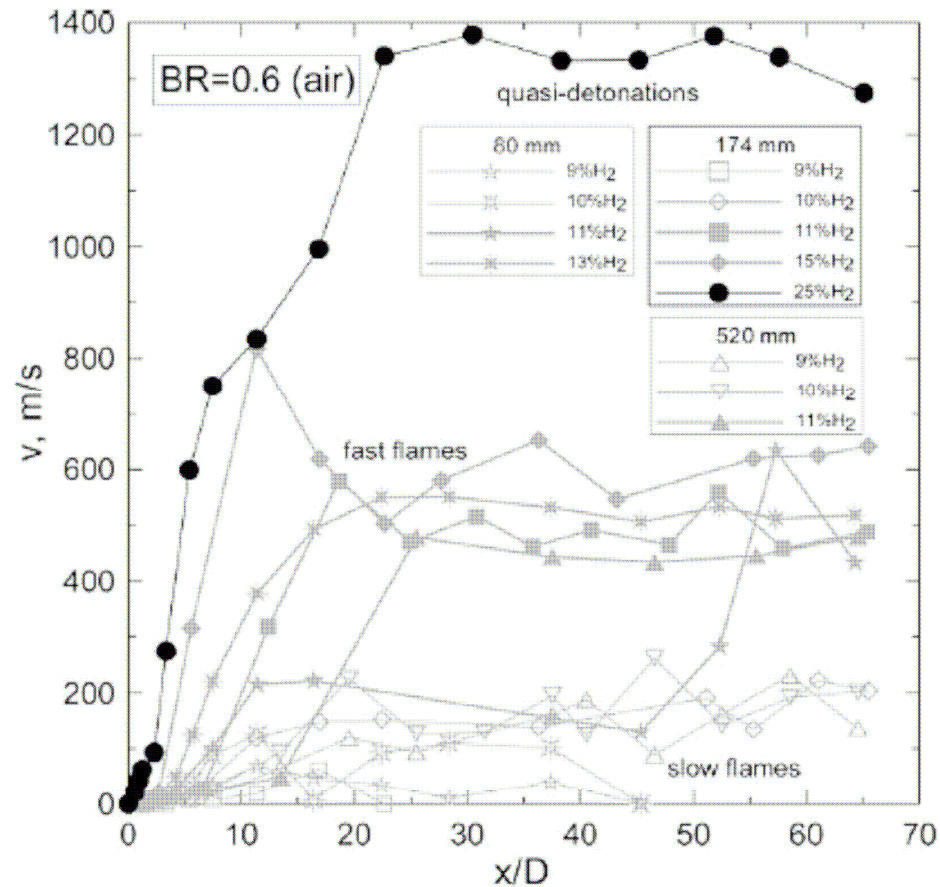
- **quenching regime** - flame fails to propagate,
- **subsonic regime** - flame is traveling at a speed that is slower than the sound speed of the combustion products,
- **choked regime (CJ Deflagration)** - flame speed is comparable with the sound speed of the combustion products,
- **quasi-detonation regime** - velocity between the sonic and Chapman-Jouguet (CJ) velocity,
- **CJ detonation regime** - velocity is equal to the CJ detonation velocity



Stoichiometric hydrogen-oxygen
 Pressure 20-150 torr
 Ignition by exploding wire

(Teodorczyk, et al., 1988)

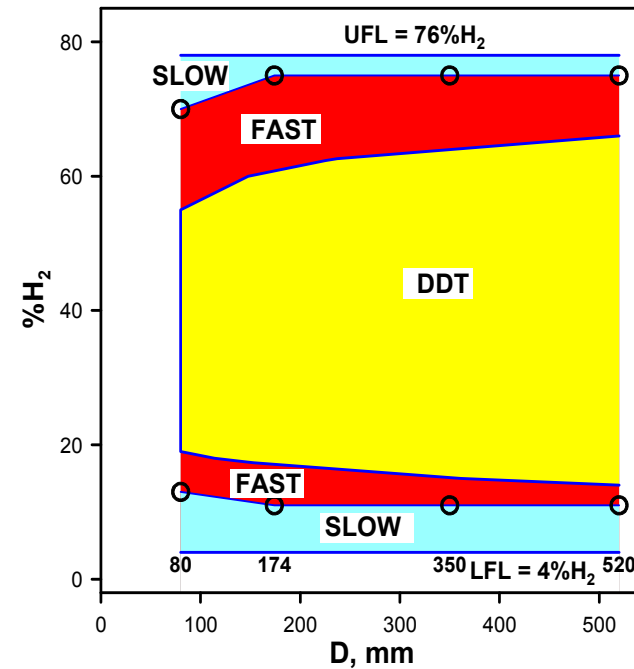
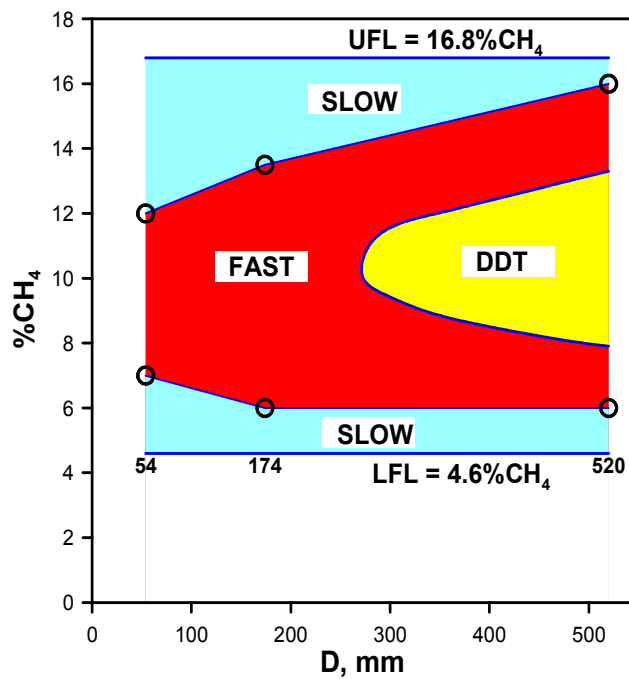
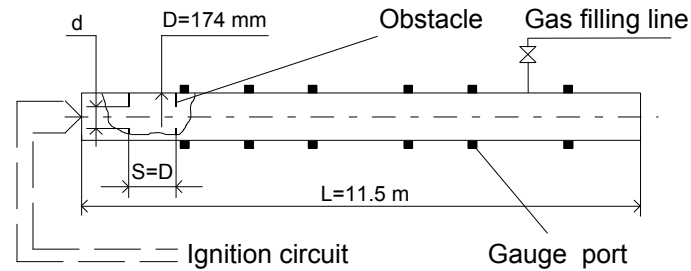




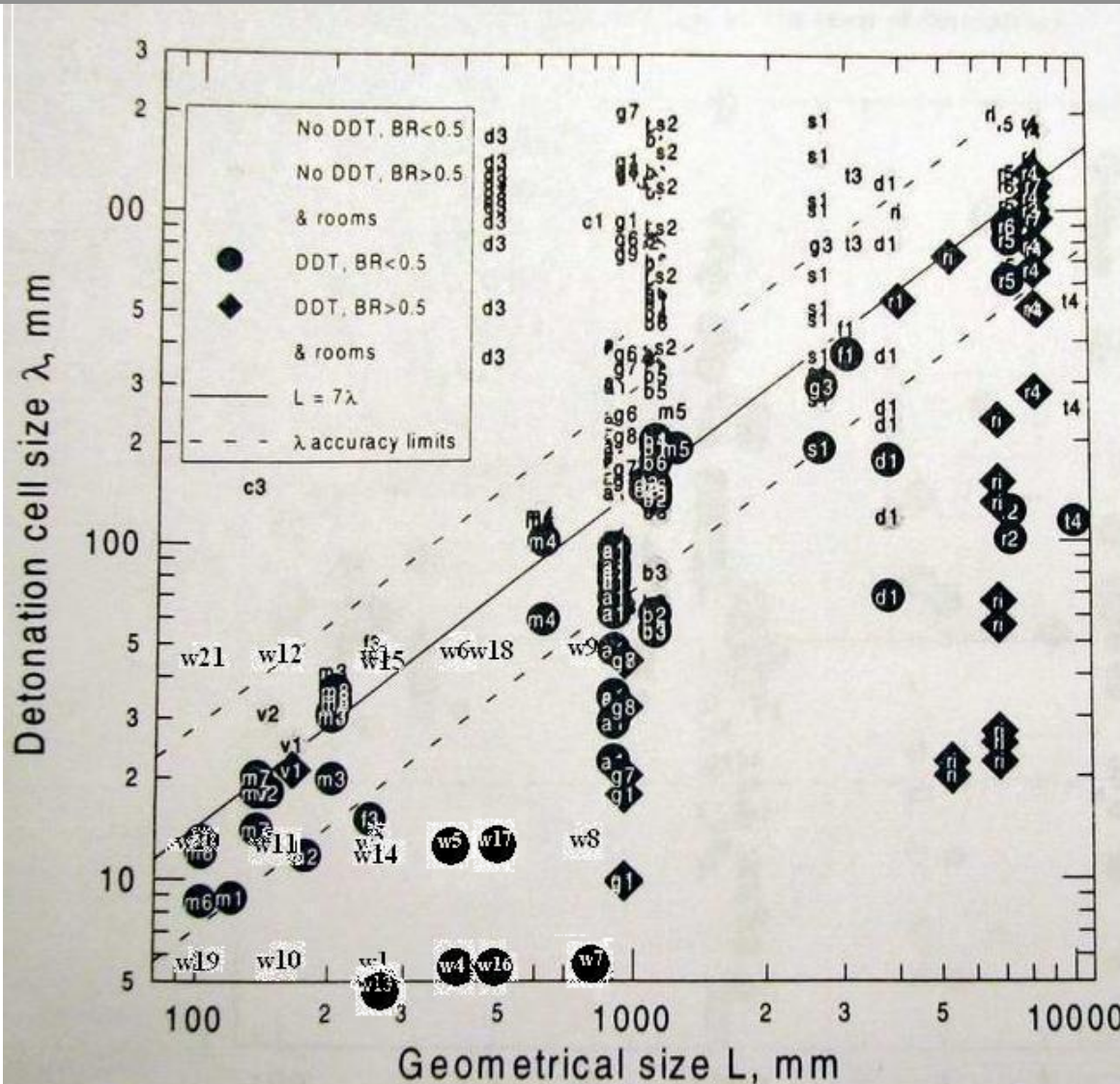
Propagation speeds of flames and detonations along the tube versus ratio of distance, x , to tube diameter, D (hydrogen–air mixtures).

Source: S.Dorofeev et al., *Journal of Loss Prevention in the Process Industries* 14 (2001) 583–589

Flame acceleration and DDT in obstructed channels



(Courtesy of M.Kuznetsov)



Geometrical size

$$L = \frac{L_1}{1 - \alpha}$$

where

$$L_1 = \frac{L + H}{2}$$

$$\alpha = \frac{1 - h}{H}$$

L – distance between obstacles

H – channel height

h – obstacle height

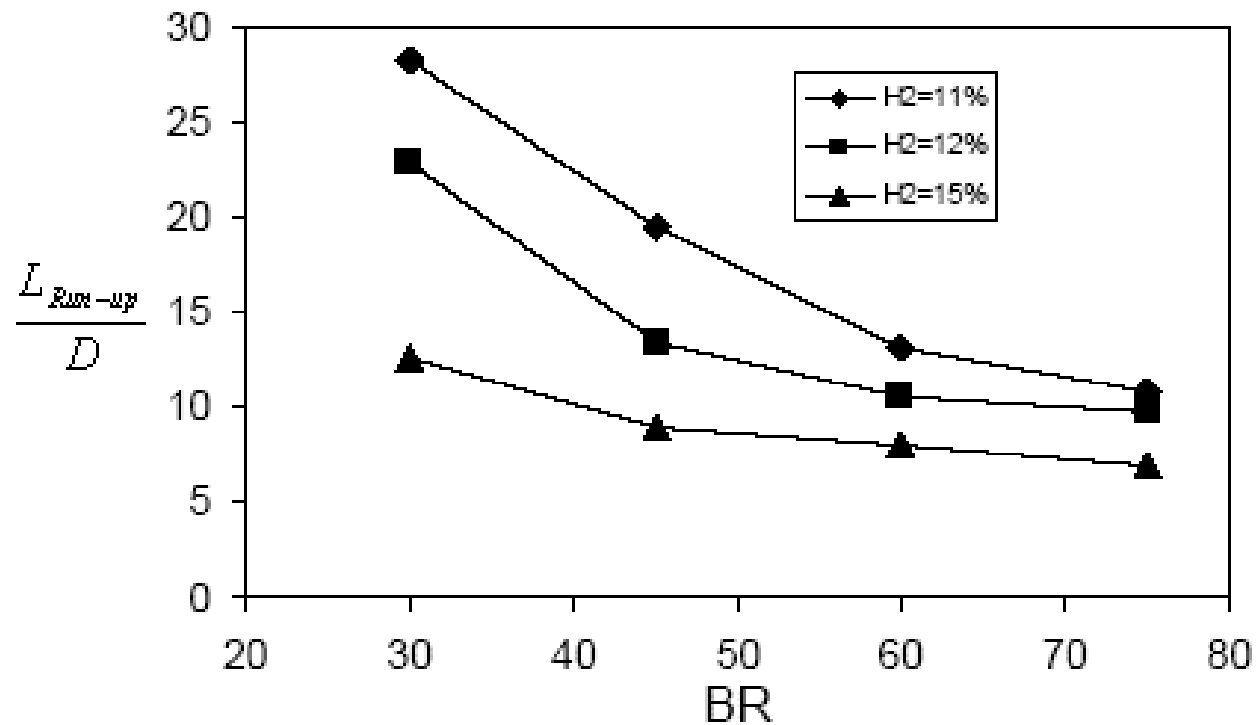
(Courtesy of S.Dorofeev)



Run-up distance for DDT in obstructed channels



In tubes at 0.1 MPa, H₂-air



(Courtesy of S.Dorofeev)

V.Gamezo et al., 31st Symposium International on Combustion, Heidelberg 2006

- stoichiometric hydrogen-air mixture at 0.1 MPa
- Reactive Navier-Stokes equations with one-step Arrhenius kinetics
- 2D channel with obstacles
- Grid: 0.02 mm (min)

